

MSFC PRACA : 2003-02-12 08:41

MSFC Problem Reporting and Corrective Action (PRACA) System
WHOLE RECORD REPORT(+ ADDENDUM)

MSFC Record # A17828	In-Flight Anomaly Number --	Contractor Report Number P-098	JSC# --	KSC# --
Problem Title REPAIR OF A 0.300 IN. LONG PENETRANT CRACK WAS NOT BOUNDED BY EXISTING WIDE PANEL DATA				
EICN# --	ELEMENT ET	Contractor LOCKHEED MARTIN	FSCM# --	FCRIT 3
HCRIT 1	Sys_Lvl --	Misc Codes A B C D E F G H I J K L M N O		
HARDWARE EIM	NOMENCLATURE --	PART# --	SER/LOT# --	MANUFACTURER --
HARDWARE LRU	NOMENCLATURE --	PART# --	SER/LOT# --	MANUFACTURER --
HARDWARE NCA	NOMENCLATURE LO2 BARREL WELD	PART# 80912400000-510	SER/LOT# 01210130	MANUFACTURER MAF
Test/Operation M - MFG	Prevailing Condition ZZ - NO PROBLEM	F / U U	Fail Mode MU - MECH TOLRNCE	Cause M - MANUFG
System PROPULSION	Defect CS - CRACK	Material S - STRUCT	Work Contact D. O'NEAL	Fail Date 01/07/2003
Received at MSFC 01/16/2003	Date Isolated 01/07/2003	FMEA Reference N/A	IFA: Mission Phase --	Mission Elapsed Time --
Location MAF		Symptom MU - MECH TOLRNCE		Time Cycle --
Effectivity Text ET-130				
Vehicle Effectivity Codes				
Vehicle 1 --	Vehicle 2 --	Vehicle 3 --	Vehicle 4 --	Vehicle 5 --
Mission Effectivity Codes				
Mssn 1 --	Mssn 2 --	Mssn 3 --	Mssn 4 --	Mssn 5 --
Estimated Completion Dates				
MSFC Approved Defer Until Date --	Contractor Req Defer Until Date --	LVL 3 Close --	Remark / Action --	
Investigation / Resolution Summary				
Last MSFC Update 01/28/2003	CN RSLV SBMT --	Defer Date --	Add Date 01/16/2003	R/C Codes 0 - EXPL -- --
Assignee				

Design M. BUTLER	Chief Engineer T. GREENWOOD	S & MA K. LAYNE	Project --	Project MGR J. SMELSER	
Approval					
Design M. BUTLER	Chief Engineer T. GREENWOOD	S & MA K.LAYNE	Project --	Project MGR J. SMELSER	
PAC Assignee T. WHITE	PAC Review Complete TLW	MSFC Closure Date 01/28/2003	Status C - CLOSED	F/A Completion --	
Problem Type --	SEV --	Program Name --	REVL --	OPRINC --	
FUNC MOD --	Software Effectivity -- - - - - -	Software Fail CD --		SUBTYPE --	Software Closure CD --
RES PERSON L2 --	Approval Signature L3 --				
Related Document Type --	Related Document ID --				
Related Document Title --					
Related Document Type --	Related Document ID --				
Related Document Title --					
Related Document Type --	Related Document ID --				
Related Document Title --					
Contractor Status Summary					
Reliability/Quality Assurance Concerns, Recommendations:					
Problem Description THE REPAIR OF A 0.300 INCH LONG PENETRANT CRACK WAS NOT BOUNDED BY EXISTING WIDE PANEL DATA DUE TO THE REPAIR ATTEMPTS REQUIRED(EIGHT), THE LOCATION OF THE DEFECT(FUSION BOUNDARY) AND THE ADJACENT NARROW WELD LAND. IN ADDITION, THE OVERALL BARREL BUILD CONSISTED OF THREE PANELS WITH NARROW WELD LANDS AND ONE SPECIAL ORDER PANEL WITH WIDER WELD LANDS.					
Contractor Investigation/Resolution 01/16/2003 - CONTRACTOR SUBMITTED OPENER/CLOSURE REQUEST AS FOLLOWS: GENERAL: ----- DURING THE BUILD OF ET 127'S LO2 BARREL, LATER REALLOCATED TO ET-130, A 0.300 INCH LONG PENETRANT CRACK WAS FOUND ALONG THE OB2 WELD FUSION BOUNDARY AT THE 84.7 - 85.0 INCH LOCATION. THE RESULTING REPAIR REQUIRED EIGHT ATTEMPTS (R8 HEAT LEVEL), PRIMARILY FROM THE OSL AND THIS REPAIR IN COMBINATION WITH THE NARROW WELD LAND OF THE ADJACENT					

PANEL, CREATED A CONDITION THAT WAS NOT BOUNDED BY EXISTING WIDE PANEL DATA. THE INITIAL WELD LAND WIDTH WAS 1.25 INCHES (DWG MIN 1.50 INCH). THE EDGE OF THE REPAIR CAME WITHIN 0.450 INCH OF THE EDGE OF THE PRIMARY WELD LAND. WIDE PANEL TESTING WAS PERFORMED AND HARDNESS DATA TAKEN FROM HARDWARE TO CONFIRM HEAT EFFECT DID NOT EXTEND INTO THE 2ND WELD LAND.

OF THE FOUR PANELS USED FOR THIS BARREL BUILD, THREE HAD BEEN CUT APART FROM OTHER PANELS FOR VARIOUS REASONS RESULTING IN REDUCED WELD LAND WIDTHS ALONG AT LEAST ONE WELD LAND ON EACH RECYCLED PANEL. THE FOURTH PANEL WAS A SPECIAL ORDER PANEL THAT HAD WIDER THAN NORMAL WELD LANDS TO ACCOMMODATE THE NARROW LANDS OF THE OTHER PANELS AND MAINTAIN THE REQUIRED BARREL CIRCUMFERENCE.

TASK I FAILURE/PROBLEM INVESTIGATION

A. EVALUATE THE ACCEPTANCE OF THIS LO2 BARREL ASSEMBLY FOR FLIGHT.

RESPONSIBILITY: G. WADGE/4420 - E. SWEET/4400
ECD: COMPLETE 10/28/02

CLOSURE STATEMENT:

OB2 WELD - WIDE PANEL TESTING AS WELL AS ANALYSIS USING STANDARD REPAIR ALLOWABLES DEMONSTRATED THE WELD REPAIR MEETS FLIGHT REQUIREMENTS, SHOWING POSITIVE MARGIN OF SAFETY (MS). AN EVALUATION OF THE PRIMARY AND SECONDARY WELD LANDS WAS ALSO PERFORMED VIA HARDNESS TESTING, THE RESULTS SHOWING MINIMAL EFFECT DUE TO REPAIR. OB1, OB3 AND OB4 WELDS - DETERMINED ACCEPTABLE FOR FLIGHT WITH NO AFFECT TO MS.

REFERENCE SENIOR MANAGEMENT REVIEW (SMR 02-005), PRESENTED 10/28/02, FOR FURTHER DETAILS.

*CAUSE:

NO CAUSE AND CORRECTIVE ACTION REQUIRED.

TASK II CORRECTIVE ACTION

NONE REQUIRED.

TASK III CLEARANCE OF EFFECTIVITIES

THIS REPAIR IS UNIQUE TO THIS ET-130. NO ADDITIONAL CLEARANCE IS REQUIRED.

TASK IV CAPS CLOSURE SUMMARY

ET-130'S LO2 BARREL CONSISTS OF THREE RECYCLED PANELS WITH ONE OR MORE NARROW WELD LANDS AND ONE SPECIAL PANEL ORDERED WITH WIDER THAN NORMAL WELD LANDS TO ACCOMMODATE THE OTHER PANELS' NARROW LANDS AND MAINTAIN THE REQUIRED BARREL CIRCUMFERENCE. DURING THE BUILD, A PENETRANT CRACK WAS FOUND ALONG A FUSION BOUNDARY OF THE OB2 WELD. DUE TO THE NARROW WELD LAND OF THE ADJACENT PANEL, THE LOCATION OF THE DEFECT AND THE EIGHT HEATS (R8) REQUIRED TO REPAIR THE DEFECT, WIDE

PANELS AND HARDNESS READINGS WERE REQUIRED. THE RESULTING ANALYSIS SHOWED THE REPAIR AND WELD ACCEPTABLE FOR FLIGHT WITH POSITIVE MS. EVALUATION OF THE OTHER THREE WELDS (OB1, OB3, OB4) FOUND NO EFFECT TO MS. BOTH A MAF SENIOR MANAGEMENT REVIEW AND MSFC SENIOR MANAGEMENT REVIEW HAVE BEEN CONDUCTED AND THIS REPAIR AND BARREL BUILD FOUND ACCEPTABLE FOR FLIGHT.

MSFC Response/Concurrence

01/28/2003 - BOARD ACCEPTED PROBLEM CLOSURE RATIONALE. SIGNATURES ARE ON FILE IN THE PAC. THIS PROBLEM IS OFFICIALLY CLOSED.

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ASSESSMENT ADDENDUM REPORT

MSFC Report# A17828	IFA# --	Contractor RPT# P-098	JSC# --	KSC# --	EICN# --
Asmnt Part# 80912400000-510	Asmnt Part Name LO2 BARREL	Asmnt Serial/Lot# 01210130			
HCRT CD 1	FCRT CD 3	CAUSE CD M - MANUFG	FAIL MODE MU - MECH TOLRNCE		
Asmnt FMEA --	Asmnt FM --	FMEA CSE --	FMEA SCSE --		
Asmnt FMEA --	Asmnt FM --	FMEA CSE --	FMEA SCSE --		
Asmnt FMEA --	Asmnt FM --	FMEA CSE --	FMEA SCSE --		
Correlated Part# --	Correlated Part# --	Correlated Part# --			
Associated LRU# --	Associated LRU# --	Associated LRU# --			
MAJOR DESIGN CHANGES					
APRV DATE --	DESCRIPTION OF CHANGES --				
ASSESSMENT TEXT					